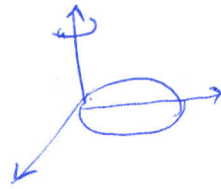
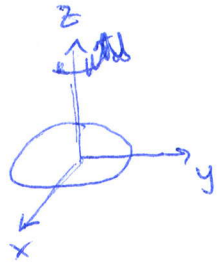


What happened here, physically?



Which moments should change?

I_{xx} , I_{zz} , but I_{yy} is physically the same!

→ parallel axis theorem / Steiner's theorem.

MOI around a given axis \vec{n} is equal to the parallel axis through the CoM plus the MOI of the ~~body~~ ^{body} ~~rotation~~ ^{w.r.t.} ~~around~~ \vec{n} , treated as a point mass" (sounds awkward)

$$\hat{I}_{ab} = \left[\int dm (r^2 \delta_{ab} - x_a x_b) \right]$$

$$= \sum_i \Delta m_i (r_i^2 \delta_{ab} - x_{ia} x_{ib}) = \sum_i \Delta m_i \left((\vec{r}_i' + \vec{R})^2 \delta_{ab} - (x_{ia}' + x_a)(x_{ib}' + x_b) \right)$$

$$= \sum_i \Delta m_i \left(\vec{r}_i'^2 \delta_{ab} + \vec{R}^2 \delta_{ab} + 2\vec{r}_i' \cdot \vec{R} \delta_{ab} - x_{ia}' x_{ib}' - x_{ia}' x_b - x_a x_{ib}' - x_a x_b \right)$$

$$= \sum_i \Delta m_i \left(\vec{r}_i'^2 \delta_{ab} - x_{ia}' x_{ib}' + \vec{R}^2 \delta_{ab} - x_a x_b + 2\vec{r}_i' \cdot \vec{R} \delta_{ab} - x_{ia}' x_b - x_a x_{ib}' \right)$$

$$= \hat{I}_{ab}' + M(\vec{R}^2 \delta_{ab} - x_a x_b) + 2M \begin{pmatrix} \vec{r}_1' \cdot \vec{R} \\ \vdots \\ 0 \end{pmatrix} \delta_{ab} - \underbrace{M x_a' x_b}_{=0} - \underbrace{M x_a x_b'}_{=0}$$

$$= \hat{I}_{ab}' + M(\vec{R}^2 \delta_{ab} - x_a x_b)$$

Now that we have

$$L = T = \frac{1}{2} \vec{\omega}^T \hat{I} \vec{\omega}$$

let's look at CoM's.

From Newton's

(will do Lagrange in ~~lect~~ lecture notes - it's a bit messier b/c Euler angles)

lab / inertial frame.

$$\frac{d\vec{p}}{dt} = \vec{F}$$

\vec{p}, \vec{L} total mom. / ang. mom.

$$\frac{d\vec{L}}{dt} = \vec{N}$$

\vec{F}, \vec{N} total force / torque

$$\vec{L} = \sum_{i=1}^A \vec{r}_i \times \vec{p}_i = \sum_{i=1}^A \Delta m_i \vec{r}_i \times \dot{\vec{r}}_i$$

body-fixed (same origin)

~~$\vec{L} = \sum \vec{r}_i \times \vec{p}_i$~~ $\leftarrow = 0$ (\vec{r}_i rotates with body)

$$\dot{\vec{r}}_i = \left(\frac{d\vec{r}_i}{dt} \right)_{BF} + \vec{\omega} \times \vec{r}_i =$$

$$\Rightarrow \vec{L} = \sum_i \Delta m_i (\vec{r}_i \times (\vec{\omega} \times \vec{r}_i)) = \sum_i \Delta m_i (r_i^2 \vec{\omega} - (\vec{r}_i \cdot \vec{\omega}) \vec{r}_i)$$

$$\Rightarrow L_a = \sum_{i=1}^A \sum_{b=1}^3 \Delta m_i (r_i^2 \delta_{ab} - x_{ib} \omega_{ia}) \omega_b = \sum_b \hat{I}_{ab} \omega_b$$

$$\Rightarrow \vec{L} = \hat{I} \vec{\omega}$$

- \vec{L} is in general not parallel to $\vec{\omega}$!!
- It will be for rotation around a principal axis (or with appropriate symmetries, e.g. for a spherical rigid body.)

Now

$$\left(\frac{d\vec{L}}{dt} \right)_{LF} = \left(\frac{d\vec{L}}{dt} \right)_{BF} + \vec{\omega} \times \vec{L}$$

$$= \hat{I} \dot{\vec{\omega}} + \vec{\omega} \times \vec{L} = \hat{I} \dot{\vec{\omega}} + \vec{\omega} \times (\hat{I} \vec{\omega}) = \vec{N}$$

\hat{I} does not depend on time in the BF fixed frame, but it will in \mathcal{L} lab frame (since rot. matrix to BF frame depends on time - dep. angles!)

Do this before you do Euler eqns

Then we have

$$\left(\dot{\vec{I}} \dot{\vec{\omega}} \right)_{BF} + \vec{\omega} \times \vec{L} = \vec{N}$$

$$\Rightarrow \vec{\omega} = \omega_{x'} \vec{e}_{x'} + \omega_{y'} \vec{e}_{y'} + \omega_{z'} \vec{e}_{z'}$$

$$\vec{L} = A \omega_{x'} \vec{e}_{x'} + B \omega_{y'} \vec{e}_{y'} + C \omega_{z'} \vec{e}_{z'}$$

(with $\vec{\omega}$ components as shown before/exercise)

$$\Rightarrow N_{x'} = A \dot{\omega}_{x'} + (C-B) \omega_{y'} \omega_{z'}$$

$$N_{y'} = B \dot{\omega}_{y'} + (A-C) \omega_{x'} \omega_{z'}$$

$$N_{z'} = C \dot{\omega}_{z'} + (B-A) \omega_{y'} \omega_{x'}$$

Euler's eqns for rigid body (in the body-fixed frame / principal axis)

General procedure:

- first determine external forces that give torque \vec{N} in lab frame
- transform to BF frame (ideally ~~is~~ principal axis system) using rotation matrices with Euler angles.

Free Top

$$\vec{N} = 0$$

→ Note that this need not mean that there are no forces! E.g. if CM fixed in lab frame

$$M \vec{R} \times \vec{g} = 0$$

$$\rightarrow A \dot{\omega}_{x'} + (C-B) \omega_{y'} \omega_{z'} = 0$$

$$B \dot{\omega}_{y'} + (A-C) \omega_{x'} \omega_{z'} = 0$$

$$C \dot{\omega}_{z'} + (B-A) \omega_{y'} \omega_{x'} = 0$$

Conserved quantities:

- multiply each equation by the w_i and add:

$$\begin{aligned}
 & A \dot{w}_x w_x + (C-B) w_y w_z w_x \\
 & + B \dot{w}_y w_y + (A-C) w_x w_z w_y \\
 & + C \dot{w}_z w_z + (B-A) w_x w_y w_z = 0 \\
 & = A \dot{w}_x w_x + B \dot{w}_y w_y + C \dot{w}_z w_z = \frac{d}{dt} \frac{1}{2} (A w_x^2 + B w_y^2 + C w_z^2) \\
 & = \frac{d}{dt} T_{rot} = 0 \quad \left(\hat{=} \text{total energy if we have no } \vec{v}_0 \text{ translation or force} \right)
 \end{aligned}$$

Multiply by $A w_x$, $B w_y$, $C w_z$:

$$\begin{aligned}
 & \frac{d}{dt} \left(A^2 \dot{w}_x w_x + A(C-B) w_y w_z w_x \right. \\
 & \quad + B^2 \dot{w}_y w_y + B(A-C) w_x w_z w_y \\
 & \quad \left. + C^2 \dot{w}_z w_z + C(B-A) w_x w_y w_z \right) = 0 \\
 & = A^2 \dot{w}_x w_x + B^2 \dot{w}_y w_y + C^2 \dot{w}_z w_z = 0 \\
 & = \frac{d}{dt} \left(\frac{1}{2} (A^2 w_x^2 + B^2 w_y^2 + C^2 w_z^2) \right) = \frac{d}{dt} \left(\frac{1}{2} \vec{L}^2 \right) \\
 & \Rightarrow \frac{d}{dt} |\vec{L}|^2 = 0
 \end{aligned}$$

$$\begin{aligned}
 \vec{L} &= \hat{I} \vec{\omega} \\
 \vec{L}^2 &= (\hat{I} \vec{\omega})^2 \\
 \vec{\omega}^T \cdot \hat{I}^T \cdot \hat{I} \vec{\omega} \\
 \hat{I}^2 &= \begin{pmatrix} A^2 & & \\ & B^2 & \\ & & C^2 \end{pmatrix}
 \end{aligned}$$

→ absolute value of \vec{L} is conserved.

To preserve \vec{L} direction:

$$A \dot{w}_x = B \dot{w}_y = C \dot{w}_z = 0$$

$$\Rightarrow (C-B) w_y w_z = 0$$

$$(A-C) w_x w_z = 0$$

$$(B-A) w_x w_y = 0$$

• If $A=B=C$ that's initially true

• If $A \neq B \neq C$: $w_x = 0, w_y = 0$

→ $\vec{L} \parallel \hat{I}^{-1} \vec{L}$ principal axis

$\vec{L} \parallel \vec{\omega}$

Stability

Let's assume rotation around z' axis, plus perturbations

$$\omega_{z'} = \omega_0 + \varepsilon_{z'}$$

$$\omega_{x'} = \varepsilon_{x'}$$

$$\omega_{y'} = \varepsilon_{y'}$$

$$\Rightarrow A \ddot{\varepsilon}_{x'} + (C-B) \varepsilon_{y'} (\omega_0 + \varepsilon_{z'}) = 0$$

$$B \ddot{\varepsilon}_{y'} + (A-C) \varepsilon_{x'} (\omega_0 + \varepsilon_{z'}) = 0$$

$$C \dot{\varepsilon}_{z'} + \underbrace{(B-A) \varepsilon_{y'} \varepsilon_{x'}}_{=0} = 0$$

$$\Rightarrow \dot{\varepsilon}_{z'} = 0$$

$$\Rightarrow \varepsilon_{z'} = \text{const.}$$

$$\Rightarrow B \ddot{\varepsilon}_{y'} + (A-C) \dot{\varepsilon}_{x'} (\omega_0 + \varepsilon_{z'}) = 0 \Rightarrow \dot{\varepsilon}_{x'} = -\frac{B}{(A-C)} \omega_0 \ddot{\varepsilon}_{y'}$$

in first eqn.

$$-\frac{AB}{(A-C)\omega_0} \ddot{\varepsilon}_{y'} + (C-B) \varepsilon_{y'} \omega_0 = 0$$

$$\ddot{\varepsilon}_{y'} - \frac{(A-C)(C-B)}{AB} \omega_0^2 \varepsilon_{y'} = 0$$

Analogously,

$$A \ddot{\varepsilon}_{x'} + (C-B) \dot{\varepsilon}_{y'} \omega_0 = 0 \Rightarrow \dot{\varepsilon}_{y'} = -\frac{A}{(C-B)\omega_0} \ddot{\varepsilon}_{x''}$$

in second eq.

$$\Rightarrow -\frac{AB}{(C-B)\omega_0} \ddot{\varepsilon}_{x''} + (A-C) \omega_0 \varepsilon_{x'} = 0$$

$$\ddot{\varepsilon}_{x''} - \frac{(A-C)(C-B)}{AB} \omega_0^2 \varepsilon_{x''} = 0$$

Define

$$\Omega^2 = \frac{(A-C)(B-C)}{AB}$$

$$\ddot{\varepsilon}_{x''} + \Omega^2 \varepsilon_{x'} = 0$$

$$\ddot{\varepsilon}_{y''} + \Omega^2 \varepsilon_{y'} = 0$$

stable solutions for $\Omega^2 > 0$,
so $A > C, B > C$ or
 $A < C$ & $B < C$ (principal
i.e., C is smallest or largest MoI

for $B < C < A$ or $A < C < B$ relation is unstable.

Intermediate Axis Theorem

Free symmetric top:

$$A=B \neq C$$

$$\Rightarrow A \ddot{\omega}_{x1} + (C-A) \omega_{y1} \omega_{z1} = 0 \quad (i)$$

$$A \ddot{\omega}_{y1} + (A-C) \omega_{x1} \omega_{z1} = 0 \quad (ii)$$

$$C \ddot{\omega}_{z1} = 0 \Rightarrow \omega_{z1} = \text{const} \equiv \omega_{0z1}$$

$$\Rightarrow A \ddot{\omega}_{x1} + (C-A) \omega_{0z1} \omega_{y1} = 0 \quad (i')$$

$$A \ddot{\omega}_{y1} + (A-C) \omega_{0z1} \omega_{x1} = 0 \quad (ii')$$

$$\Rightarrow A \ddot{\omega}_{x1} + (C-A) \omega_{0z1} \omega_{y1} = 0 \Rightarrow \ddot{\omega}_{y1} = \frac{-A}{(C-A) \omega_0} \ddot{\omega}_{x1}$$

in (ii')

~~$$\frac{A^2 \ddot{\omega}_{y1}}{(C-A) \omega_0} + (A-C) \omega_{0z1} \omega_{x1} = 0$$~~

$$\frac{A^2}{(A-C) \omega_0} \ddot{\omega}_{x1} + (A-C) \omega_{0z1} \omega_{x1} = 0$$

$$\ddot{\omega}_{x1} + \frac{(A-C)^2 \omega_{0z1}^2}{A^2} \omega_{x1} = 0$$

Analogously

$$\ddot{\omega}_{y1} + \frac{(C-A)^2}{A^2} \omega_{0z1}^2 \omega_{y1} = 0$$

$$\equiv \Omega^2$$

$$\Rightarrow \omega_{x1}(t) = \omega_{0z1} \sin(\Omega t + \beta)$$

$$\omega_{y1}(t) = \omega_{0z1} \cos(\Omega t + \beta)$$

$\Rightarrow \omega_{0z1}$ is const.

$$\omega = \sqrt{\omega_{0z1}^2 + \omega_{0z1}^2} = \text{const.}$$

$\vec{\omega}_\perp(t) = (\omega_{x1}(t), \omega_{y1}(t))$ is a circle in the $\overset{BF}{xy}$ plane

ω_{0z1} ?

can show -
same as
intermediate
axis

since ω_x and ω_y
are related.

ω_{0z1}, β must be
the same